



**[4910-13-P]**

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 39**

**[Docket No. FAA-2014-0832; Directorate Identifier 2014-SW-044-AD; Amendment 39-17995; AD 2014-21-03]**

**RIN 2120-AA64**

**Airworthiness Directives; Airbus Helicopters (Previously Eurocopter France)**

**Helicopters**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule; request for comments.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for Airbus Helicopters Model AS332L2 helicopters with a certain yaw control damper support (support) installed. This AD requires repetitively inspecting the support attachment points for a crack. This AD is prompted by a report that the front attachment points of several supports were found to have cracks. These actions are intended to detect a crack in a support, which could result in failure of the support, separation of the yaw damper unit, blocking of the yaw flight control channel, and reduced control of the helicopter.

**DATES:** This AD becomes effective [INSERT DATE 15 DAYS AFTER DATE OF PUBLICATION IN THE FEDERAL REGISTER].

We must receive comments on this AD by [INSERT DATE 60 DAYS AFTER DATE OF PUBLICATION IN THE FEDERAL REGISTER].

**ADDRESSES:** You may send comments by any of the following methods:

- Federal eRulemaking Docket: Go to <http://www.regulations.gov>. Follow the online instructions for sending your comments electronically.

- Fax: 202-493-2251.

- Mail: Send comments to the U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590-0001.

- Hand Delivery: Deliver to the “Mail” address between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

### **Examining the AD Docket**

You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the Docket Operations Office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the European Aviation Safety Agency (EASA) AD, any comments received, and other information. The street address for the Docket Operations Office (telephone 800- 647-5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

For service information identified in this AD, contact Airbus Helicopters, Inc., 2701 N. Forum Drive, Grand Prairie, TX 75052; telephone (972) 641-0000 or (800) 232-0323; fax (972) 641-3775; or at <http://www.airbushelicopters.com/techpub>. You may review the referenced service information at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137.

**FOR FURTHER INFORMATION CONTACT:** Rao Edupuganti, Aviation Safety Engineer, Regulations and Policy Group, Rotorcraft Directorate, FAA, 2601 Meacham

Blvd., Fort Worth, Texas 76137; telephone (817) 222-5110; email  
rao.edupuganti@faa.gov.

## **SUPPLEMENTARY INFORMATION:**

### **Comments Invited**

This AD is a final rule that involves requirements affecting flight safety, and we did not provide you with notice and an opportunity to provide your comments prior to it becoming effective. However, we invite you to participate in this rulemaking by submitting written comments, data, or views. We also invite comments relating to the economic, environmental, energy, or federalism impacts that resulted from adopting this AD. The most helpful comments reference a specific portion of the AD, explain the reason for any recommended change, and include supporting data. To ensure the docket does not contain duplicate comments, commenters should send only one copy of written comments, or if comments are filed electronically, commenters should submit them only one time. We will file in the docket all comments that we receive, as well as a report summarizing each substantive public contact with FAA personnel concerning this rulemaking during the comment period. We will consider all the comments we receive and may conduct additional rulemaking based on those comments.

### **Discussion**

This AD is prompted by AD No. 2014-0080, dated March 27, 2014, issued by EASA, which is the Technical Agent for the Member States of the European Union, to correct an unsafe condition for Airbus Helicopters Model AS332L2 helicopters with a support part number (P/N) 332A25-1334-00 installed. EASA advises of several reports of cracks on the two front attachment points of the support, and that subsequent

investigations determined pilot actions on the yaw pedals could generate detrimental loading conditions on the support attachment points and initiate a crack. EASA also states that a crack could lead to structural failure of the support, detachment of the damper unit, possible blocking of the yaw flight control channel, and reduced control of the helicopter. EASA AD No. 2014-0080 requires repetitive inspections of the support and, if there is a crack, replacing the support.

#### **FAA's Determination**

These helicopters have been approved by the aviation authority of France and are approved for operation in the United States. Pursuant to our bilateral agreement with France, EASA, its technical representative, has notified us of the unsafe condition described in the EASA AD. We are issuing this AD because we evaluated all information provided by EASA and determined the unsafe condition exists and is likely to exist or develop on other helicopters of the same type design.

#### **Related Service Information**

We reviewed Airbus Helicopters Alert Service Bulletin No. AS332-05.00.98, Revision 0, dated March 26, 2014 (ASB), for Model AS332L2 helicopters. This ASB describes procedures for inspecting the support attachment points for a crack and, if there is a crack, requires replacing the support.

#### **AD Requirements**

This AD requires, for helicopters with 3,900 or more hours time-in-service (TIS), within 100 hours TIS and at intervals not exceeding 825 hours TIS, repetitively inspecting each support, P/N 332-A25-1334-00, at its attachment points for a crack. If there is a crack, this AD requires replacing the support before further flight.

**Interim Action**

We consider this AD to be an interim action. If final action is later identified, we might consider further rulemaking then.

**Costs of Compliance**

There are no costs of compliance with this AD because there are no helicopters with this type certificate on the U.S. Registry.

**FAA's Justification and Determination of the Effective Date**

There are no helicopters with this type certificate on the U.S. Registry. Therefore, we believe it is unlikely that we will receive any adverse comments or useful information about this AD from U.S. Operators.

Since an unsafe condition exists that requires the immediate adoption of this AD, we determined that notice and opportunity for public comment before issuing this AD are unnecessary because there are none of these products on the U.S. Registry.

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority

because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

### **Regulatory Findings**

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed, I certify that this AD:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska to the extent that it justifies making a regulatory distinction; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

### **List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### **Adoption of the Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

### **PART 39 - AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

### **§ 39.13 [Amended]**

2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

2014-21-03 **Airbus Helicopters (Previously Eurocopter France):** Amendment 39-17995; Docket No. FAA-2014-0832; Directorate Identifier 2014-SW-044-AD.

#### **(a) Applicability**

This AD applies to Airbus Helicopters Model AS332L2 helicopters with a yaw control damper support (support) part number 332A25-1334-00 installed, certificated in any category.

#### **(b) Unsafe Condition**

This AD defines the unsafe condition as a crack on a support at an attachment point, which could result in failure of the support, separation of the yaw damper unit, blocking of the yaw flight control channel, and reduced control of the helicopter.

#### **(c) Effective Date**

This AD becomes effective [INSERT DATE 15 DAYS AFTER DATE OF PUBLICATION IN THE FEDERAL REGISTER].

#### **(d) Compliance**

You are responsible for performing each action required by this AD within the specified compliance time unless it has already been accomplished prior to that time.

#### **(e) Required Actions**

For helicopters with 3,900 hours time-in-service (TIS) or more, within 100 hours TIS and thereafter at intervals not exceeding 825 hours TIS, using a light source and a mirror, inspect each support at the four attachment points for a crack. If there is a crack,

before further flight, replace the support.

**(f) Special Flight Permits**

Special flight permits are prohibited.

**(g) Alternative Methods of Compliance (AMOCs)**

(1) The Manager, Safety Management Group, FAA, may approve AMOCs for this AD. Send your proposal to: Rao Edupuganti, Aviation Safety Engineer, Regulations and Policy Group, Rotorcraft Directorate, FAA, 2601 Meacham Blvd., Fort Worth, Texas 76137; telephone (817) 222-5110; email [rao.edupuganti@faa.gov](mailto:rao.edupuganti@faa.gov).

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, we suggest that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office, before operating any aircraft complying with this AD through an AMOC.

**(h) Additional Information**

(1) Airbus Helicopters Alert Service Bulletin No. AS332-05.00.98, Revision 0, dated March 26, 2014, which is not incorporated by reference, contains additional information about the subject of this AD. For service information identified in this AD, contact Airbus Helicopters, Inc., 2701 N. Forum Drive, Grand Prairie, TX 75052; telephone (972) 641-0000 or (800) 232-0323; fax (972) 641-3775; or at <http://www.airbushelicopters.com/techpub>. You may review a copy of the service information at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth Texas 76137.



(2) The subject of this AD is addressed in European Aviation Safety Agency (EASA) AD No. 2014-0080, dated March 27, 2014. You may view the EASA AD on the Internet at <http://www.regulations.gov> in Docket No. FAA-2014-0832.

**(i) Subject**

Joint Aircraft Service Component (JASC) Code: 6700: Rotorcraft Flight Control.  
Issued in Fort Worth, Texas, on October 6, 2014.

Lance T. Gant,  
Acting Manager, Rotorcraft Directorate,  
Aircraft Certification Service.

[FR Doc. 2014-25406 Filed 10/24/2014 at 8:45 am; Publication Date: 10/27/2014]